



Division of Strength of Materials and Structures

Faculty of Power and Aeronautical Engineering

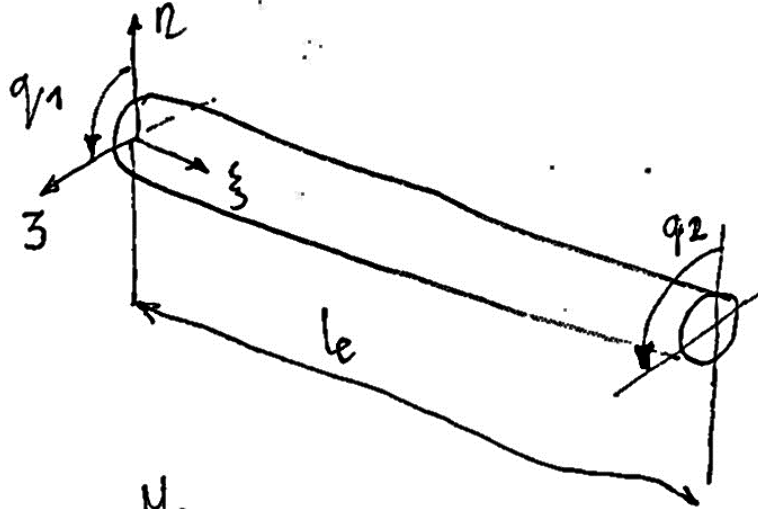


Finite element method (FEM1)

Lecture 8B. Torsion bar finite element

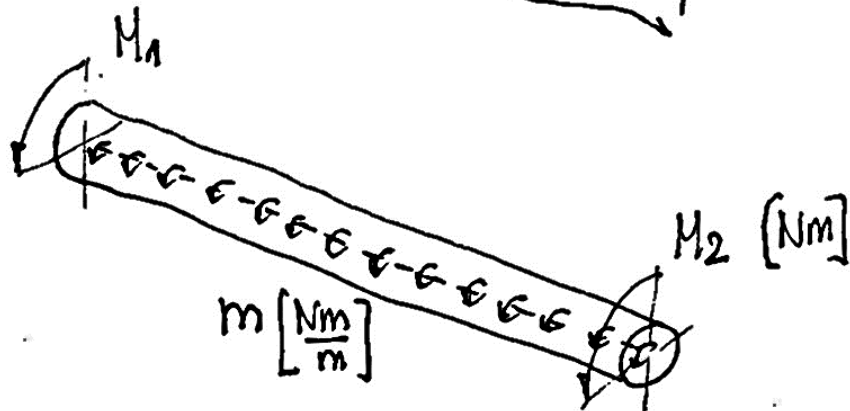
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Finite element of a rod loaded with a torsional moment



q_1, q_2 - rotations
(twist angles)

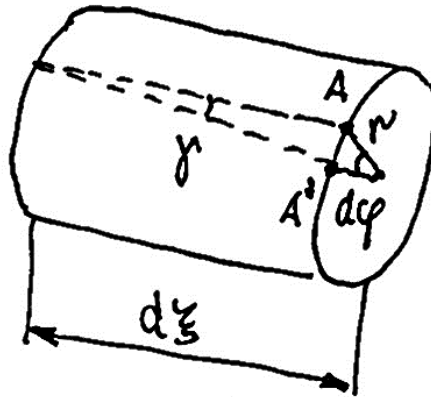
Load:



M - torque

Torque per unit length

Torsion bar finite element - deformations



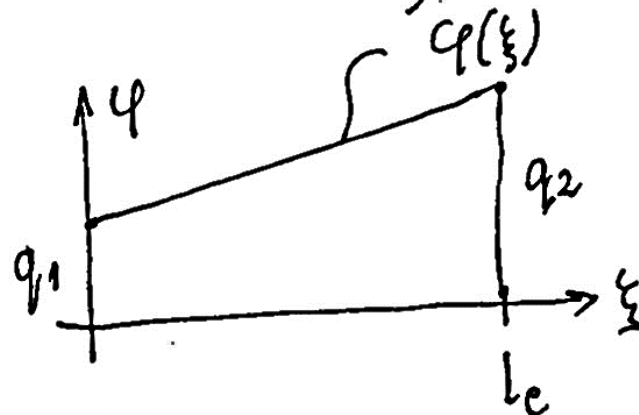
Shape functions:

$$N_1(\xi) = 1 - \frac{\xi}{l_e}$$

$$N_2(\xi) = \frac{\xi}{l_e}$$

$$AA' \cong \gamma \cdot d\xi = r d\varphi$$

$$\gamma = r \frac{d\varphi}{d\xi}$$



$$\varphi(\xi) = N_1(\xi) \cdot q_1 + N_2(\xi) \cdot q_2 =$$

$$= \underset{1 \times 2}{[N]} \cdot \underset{2 \times 1}{\{q\}}_e$$

$$\frac{d\varphi}{d\xi} = \underset{2 \times 1}{\left[\frac{dN_1}{d\xi}, \frac{dN_2}{d\xi} \right]} \cdot \underset{2 \times 1}{\{q\}}_e = \underset{2 \times 1}{\left[-\frac{1}{l_e}, \frac{1}{l_e} \right]} \cdot \{q\}_e$$

Torsion bar finite element – elastic strain energy

$$U_e = \frac{1}{2} \int_{\Omega_e} \underbrace{\tau}_{\uparrow} \cdot \gamma \, d\Omega_e = \frac{1}{2} \int_{\Omega_e} G \cdot \gamma^2 \, d\Omega_e = \frac{1}{2} \int_{\Omega_e} G r^2 \left(\frac{d\varphi}{d\xi} \right)^2 d\Omega_e =$$

Hook's law:

$$\tau = G \cdot \gamma = \frac{E}{2(1+\nu)} \cdot \gamma$$

$$= \frac{1}{2} \int_0^l G \frac{d\varphi}{d\xi} \cdot \frac{d\varphi}{d\xi} \cdot \underbrace{\int_A r^2 dA}_{\text{polar moment of inertia } J_s} d\xi =$$

polar moment of inertia J_s

$$J_s = \frac{\pi d^4}{32}$$

For a circular cross section

Torsion bar finite element – stiffness matrix

$$= \frac{1}{2} \int_0^l GJ_s \cdot \frac{d\varphi}{d\xi} \cdot \frac{d\varphi}{d\xi} d\xi =$$

$\begin{matrix} \uparrow & & \uparrow \\ Lq \rfloor_e \cdot \left\{ \begin{matrix} \frac{dN_1}{d\xi} \\ \frac{dN_2}{d\xi} \end{matrix} \right\} & & \left[\frac{dN_1}{d\xi}, \frac{dN_2}{d\xi} \right] \cdot \left\{ q \right\}_e \end{matrix}$

 $\begin{matrix} 1 \times 2 & & 2 \times 1 \end{matrix}$

$$= \frac{1}{2} Lq \rfloor_e \cdot [k]_e \cdot \left\{ q \right\}_e ; \quad \text{where:}$$

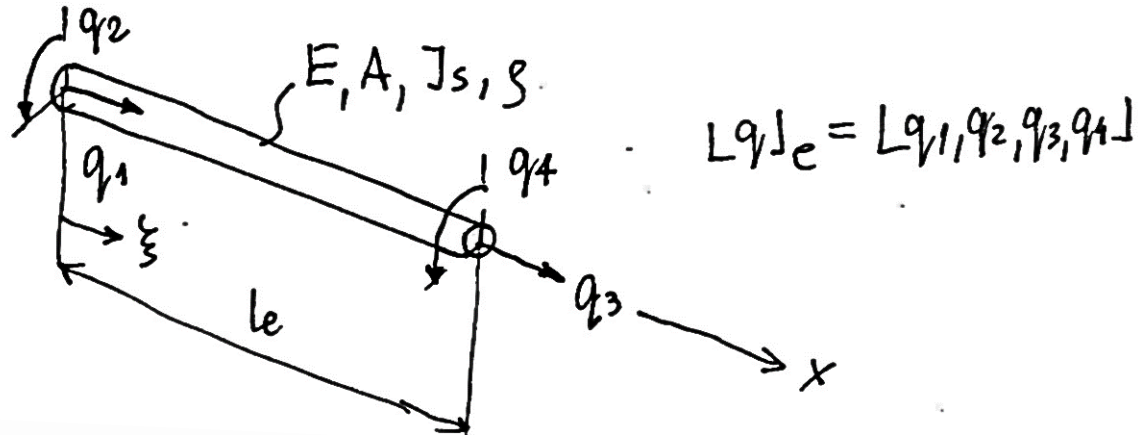
$$[k]_e = GJ_s \begin{bmatrix} \int_0^l \frac{dN_1}{d\xi} \frac{dN_1}{d\xi} d\xi & \int_0^l \frac{dN_1}{d\xi} \frac{dN_2}{d\xi} d\xi \\ \int_0^l \frac{dN_2}{d\xi} \frac{dN_1}{d\xi} d\xi & \int_0^l \frac{dN_2}{d\xi} \frac{dN_2}{d\xi} d\xi \end{bmatrix} = \frac{GJ_s}{l} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

Torsion bar finite element – potential energy of the load

$$W_e = \underset{1 \times 2}{[F]}_e \cdot \{q\}_e + M_1 \cdot q_1 + M_2 \cdot q_2$$

$$\underset{1 \times 2}{[F]}_e = \left[\int_0^L m(\xi) \cdot N_1(\xi) d\xi, \int_0^L m(\xi) \cdot N_2(\xi) d\xi \right]$$

Element loaded with axial force and torsional moment



Element loaded with axial force:

Element loaded with a torsional moment:

$$[k]_A = \frac{EA}{l_e} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

2×2

$$[q]_A = [q_1, q_3] \quad (\text{translational D.O.F})$$

$$[k]_T = \frac{GJ_s}{l_e} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

2×2

$$[q]_T = [q_2, q_4] \quad (\text{rotational D.O.F})$$

$$[k]_A^* = \frac{EA}{l_e} \begin{bmatrix} 1 & 0 & -1 & 0 \\ 0 & 0 & 0 & 0 \\ -1 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix}$$

4×4

$$[k]_T^* = \frac{GJ_s}{l_e} \begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & -1 \\ 0 & 0 & 0 & 0 \\ 0 & -1 & 0 & 1 \end{bmatrix}$$

4×4

Element loaded with axial force and torsional moment – stiffness matrix

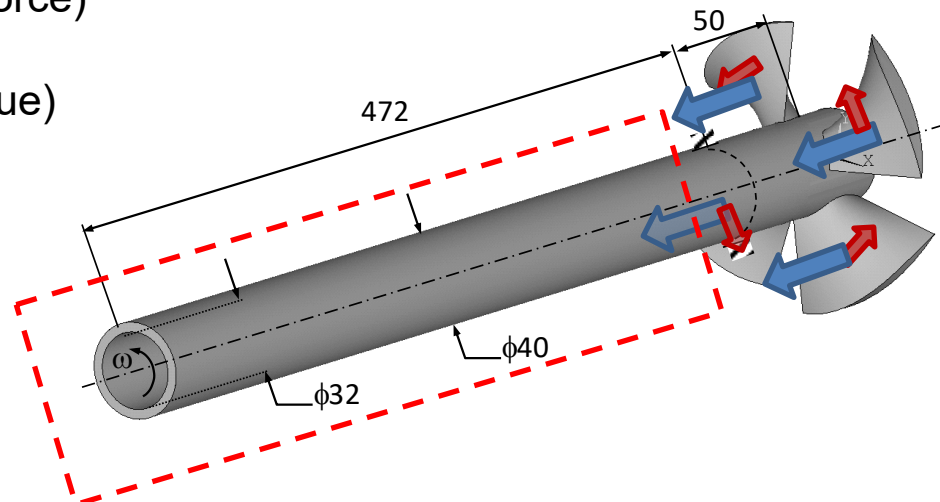
$$[k]_e = [k]_A^* + [k]_T^* = \frac{1}{l_e} \begin{bmatrix} EA & 0 & -EA & 0 \\ 0 & GJ_s & 0 & -GJ_s \\ -EA & 0 & EA & 0 \\ 0 & -GJ_s & 0 & GJ_s \end{bmatrix}$$

4×4 4×4 4×4

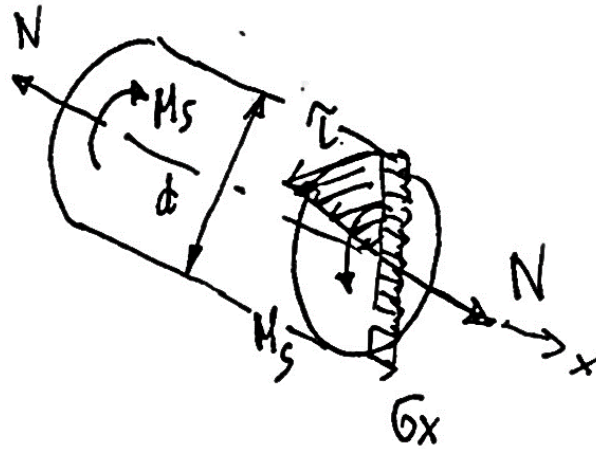
Example: An element loaded with an axial force and a torsional moment used in the FEM model of a ship's propeller shaft.

← lift force (axial force)

← drag force (torque)



Stress state in the FEM model of the ship's propulsion shaft



$$\begin{aligned}\sigma_x &= E \cdot \varepsilon_x = E \cdot \frac{du}{d\xi} = \\ &= E \left[\frac{dN_1}{d\xi}, \frac{dN_2}{d\xi} \right] \cdot \begin{Bmatrix} q_1 \\ q_3 \end{Bmatrix}_A = \\ &= E \frac{q_3 - q_1}{l_e}\end{aligned}$$

$$\frac{dN_1}{d\xi} = -\frac{1}{l_e} \quad , \quad \frac{dN_2}{d\xi} = \frac{1}{l_e}$$

$$\begin{aligned}\tau(r) &= G \cdot \gamma(r) = G \cdot r \frac{d\varphi}{d\xi} = G \cdot r \left[\frac{dN_1}{d\xi}, \frac{dN_2}{d\xi} \right] \cdot \begin{Bmatrix} q_2 \\ q_4 \end{Bmatrix}_T = \\ &= G \cdot \frac{q_4 - q_2}{l_e} \cdot r \quad , \quad \tau_{\max} = \frac{Gd}{2} \cdot \frac{q_4 - q_2}{l_e}\end{aligned}$$

$$\sigma_{eqv}^{\max} = \sqrt{\sigma_x^2 + 3\tau_{\max}^2}$$

